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(54) A propellant charge for cannons and a method of producing such a charge.

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Description**TECHNICAL FIELD**

The present invention relates to a propellant charge for cannons which may be given high charge density and high progressivity, and also to a method of producing such a charge.

BACKGROUND ART

As a rule, propellant charges for cannons consist either of loose grains or flakes, rods or perforated cylinders, or of sticks with or without perforations enclosed in a case or cartridge, in which the length of the perforated propellant sticks has, for combustion reasons, hitherto been generally limited to a maximum of 100 times the diameter of the powder's inner-channels. Moreover, charges of this type are often composed of mixtures of different batches of two or more powders. In this context, it is of minor importance whether the powder is - from the purely chemical point of view of the single, double or varying triple-base type or whether or not the surface is coated or inhibited.

Hence, the present invention now relates to propellant charges for cannon ammunition which may be given an extremely high charge density and high progressivity, the invention also relating to a method of producing such charges.

Generally speaking, a normal propellant may be considered as having a specific gravity of approx 1.53. In charges consisting of propellant of loose grains or sticks of the above-mentioned types, a charge density of approx 0.9 g/cc will be attained. On the other hand, according to the present invention, it will be possible to produce charges with charge densities of up to 1.4 g/cc i.e. a considerable increase. This may be of value in those cases where the strength parameters and performance of a cannon, for example a howitzer, a tank gun or a naval gun, enable the weapon to withstand larger propellant charges than could be accommodated in the chamber of the weapon if the powder were in the loose state or in sticks. Thus, the charge according to the present invention would make it possible to improve the performance of older high-quality artillery pieces.

The present invention thus relates to a method of producing propellant charges for cannons and to charges produced according to said method, said charges having a high charge density and a high progressivity achieved by a combination of loosely added disoriented propellant powder in the form of flakes, powder grains, short powder rods or tubes and formed propellant powder in the form of densely packed single or multi-hole tubular propellant sticks with interior combustion channels and of

long length in the relation to said combustion channels and with a total length corresponding to at least the major fraction of the available charge space for the charge in consideration.

Propellant charges of high progressivity comprising as well loosely added disoriented propellant grains as formed propellant is however described in EP-A-0082758 which document forms the basis for the pre-characterising parts of claims 1 and 3. The formed propellant according to said specification is however added as fragmentable balls of propellant with a diameter smaller than the orifice of the charge in consideration.

The long single- or multi-hole propellant sticks which are part of the charge according to the invention have however also prior to the addition to the charge according to the invention at predetermined separations been provided with perforations from the outside of the propellant stick reaching all of its combustion channels.

Propellant charges of densely packed multi-hole propellant rods of long length when compared to the diameter of the combustion channels provided with perforations at pre-determined separations are however known per se through DE-C-135102. The perforations according to said patent have however been carried out in the form of open cuts achieved by cutting off some part of the propellant material and this will make said perforations function as gasoutlets when the propellant rods are ignited at their outside and along the combustion channels. The gasvelocity in such gasoutlets will be very much increased which will erode the area of the gasoutlet and thereby cause an incomplete combustion of the propellant material.

According to the present invention this drawback has however been avoided by effecting each of said perforations transversally through the propellant tubes and without the removal of any propellant material and of sufficient width to pass through all of the combustion channels in the propellant tube. The features of the invention are described in the appended claims.

The advantage inherent in such weakened portions is that the sticks will, during the initial phase of the combustion, be shattered transversally across these weakened points by the internal gas pressure, whereafter combustion of this part of the charge will be effected in the same manner as a charge consisting initially of perforated propellant grains of short lengths.

Single- or multi-hole tubular propellant rods provided with this particular type of perforations effected transversally through the propellant rod and without the removal of any propellant material and of a sufficient width to pass through all of the combustion channels in the propellant tube are

described in our co-pending European patent 0 304 100.

Another aspect of the present invention relates to the grain, flake-, tube or rod-shaped powder which fills out the remaining portion of the available charge space. This amount of loose powder may, if desired, be compacted around the bundle or bundles of propellant sticks. According to the present invention, the propellant stick bundles suitably consist of 1-, 7-, 19- or 37-hole perforated powder of optimal outer configuration. Other forms and numbers of perforations may also come into consideration.

A propellant charge of the type according to the present invention may be given an extremely high charge density, in that a considerable part of its total volume consists of densely packed tubular propellant sticks. If, moreover, these sticks are exteriorly inhibited by a substance of lower burning speed than the propellant, this part of the charge will obtain a very high progressivity in that the propellant sticks will, at least initially, be burned from the inside with a consequently successively increasing burning area. When the inhibiting substance has been consumed there will, moreover, be obtained a large stepwise increase in the burning area which makes for further increased gas generation. The previously mentioned weakenings of the propellant sticks are suitably applied at separations of between 10 and 100 times the diameter of the inner burning channels of the tubular propellant. The burning channels of the tubular propellant should preferably be connected with at least every second one of these weakened points.

As a result of the weakened points, there will be obtained a rod-shaped tubular propellant which, in terms of handling and during the ignition phase, will function as a tubular propellant of full length, but which, during burning becomes shattered and is ultimately totally combusted as though it consisted of a normal tubular propellant cut into short lengths. Otherwise, tubular propellant of long lengths shows a marked tendency, after a brief period of combustion, to be shattered by the internal gas pressure into irregular fragments which readily give rise to disastrous pressure peaks in the barrel.

This problem is wholly obviated according to the present invention. One type of weakening which has been established as being particularly advantageous comprises a symmetrically disposed perforation of determined width and completely executed without the removal of any propellant. Thus, such a perforation rather assumes the form of a central incision. Since the perforation leaves a certain amount of propellant on either side, the tubes may retain nevertheless a high degree of cohesion and stability. Moreover, in multi-hole propellant, it is

often possible to cover, with one single incision, all combustion channels, which ensures a rapid internal overall ignition, effective cracking zones and high charge weight while still imparting superior initial cohesion to the tubes.

As was mentioned above, the weakened propellant sticks will be burnt in the same manner as a corresponding amount of tubular propellant originally cut into shorter pieces, apart from the fact that this burning takes place within an overall smaller volume, since the propellant sticks according to the present invention are, from the outset, concentrated within a smaller volume than would have been the case in loosely disposed, shorter tubular propellant rods.

Another way to prevent the very long propellant sticks from breaking in an uncontrolled manner during combustion is thus to provide every burning channel with a longitudinal slit through to the outside surface, but the propellant will, as already stated, then become degressive.

The primary task of that portion of the charge according to the present invention which consists, from the outset, of loosely disposed, i.e. disorientated flakes, powder grains or short powder rods or tubes lying around the aligned sticks and which may be compacted within reasonable limits is to ensure a rapid total ignition of the entire charge and a rapid initial gas generation. It is also a simple matter to use loose powder to fill out those parts of a throated case which may not be filled, without difficulty, with densely packed bundles of propellant sticks.

The mixture of weakened tubular propellant and one or more types of loosely disposed powder provides moreover considerable opportunities for controlling the combustion process at the desired pressure vs time cycle in the barrel.

The simplest method of filling a case with a propellant charge according to the present invention is based on the procedure of first arranging one or more bundles of long propellant sticks whose total outer diameter generally corresponds to the inner diameter of the throat of the case, or other desired outer dimension, whereafter the bundle, or the bundles, is passed down into the case and the remaining space within the case is filled with loose powder.

It is also conceivable according to the present invention to dispose tubular bundles of propellant sticks such that the loose powder is filled in the middle of the charge. Furthermore, a certain portion, for example the rear half of a cartridge case, may be filled with densely packed weakened tubular propellant sticks and the remainder with powder grains or flakes of any optional type.

BRIEF DESCRIPTION OF THE ACCOMPANYING DRAWINGS

The present invention has been defined in the appended claims and will now be described in greater detail in conjunction with the accompanying drawings:

In the accompanying drawings:

Fig. 1 shows a portion of a propellant stick weakened perpendicular to the longitudinal axis;

Fig. 2 shows, on a larger scale, a section through one such example of weakened rosette-shaped propellant stick;

Fig. 3 is an end elevation of densely packed propellant sticks; and

Fig. 4 shows, on a smaller scale, a longitudinal projection partially in section, of a case charged according to the present invention.

DESCRIPTION OF PREFERRED EMBODIMENT

Referring to the drawings, the propellant stick or strand 1 shown in Fig. 1 consists of a seven hole perforated rosette-shaped propellant stick weakened by perforations whose cross-section (on a larger scale) will be apparent from Fig. 2. Thus, the propellant stick 1 is provided with longitudinal burning channels consisting of six edge channels 2 and one centre channel 3, the stick being further provided with a number of through-perforations 4.

The distance between two weakened points, in the present context perforations of the propellant stick, is designated a in Fig. 1. This distance corresponds to a suitable length for a tubular propellant. The reason for this is that the length should not be too great, since otherwise it could give rise to critical gas speeds near the discharge mouths of the channels. As soon as gas generation has become initiated, the propellant sticks according to the present invention will be shattered at the weakened points. These thus ensure a complete total ignition along the channels 2 and 3 and serve as fracture zones when the gas pressure rises.

Fig. 2 illustrates a particularly advantageous method of weakening, for example, rosette-shaped sticks, shown in the figure as a 7-hole rosette-shaped stick weakened by means of a centrally placed total perforation 4 without the removal of any propellant, in which each perforation 4 passes through all of the burning channels 2 and 3 of the propellant but leaves a sufficient amount of propellant 5 on either side of the perforation in order for the propellant to retain good stability. The propellant stick illustrated in Fig. 1 is treated in this manner.

As will be apparent from Fig. 3, propellant sticks 12 split in the longitudinal direction may be used as filler along the periphery of the propellant

stick bundles. Special bands for retaining the bundled configuration of the propellant sticks are designated by 6 in Fig. 4. Fig. 4 also shows a case 7 with its associated primer 8; 9 relates to loosely packed, disorientated propellant grains in the form of flakes, rods or tubes of short length.

Thus, in the alternative illustrated in Fig. 4, the charge consists, on the one hand, of a bundle of tubular sticks (11) weakened in the manner described above and/or possessing longitudinal slits, whose total outer diameter is not greater than the inner diameter of the neck of the case, and, on the other hand, of loosely packed propellant 9. In this case, the propellant stick bundle is of full length, i.e. it extends from the bottom of the case to the lowest position 10 of the projectile proper in the case. However, it is also conceivable that the charge may be divided up into several sub-charges. The loose propellant 9 may possibly be slightly compacted. As has been mentioned in the foregoing, any number of other combinations between a bundle of weakened or longitudinally slotted, tubular propellant sticks and loosely packed propellant may be employed for producing charges which give the desired pressure vs time cycle in every individual case.

Claims

- 30 1. A method of producing propellant charges for cannons with high charge density and high progressivity comprising within a charge space in combination formed propellant powder (1) and loosely added disoriented propellant powder (11) which in the form of flakes, powder grains, short single- or multi-hole powder rods or tubes (9) takes up the remaining available charge space not occupied by the formed propellant powder characterized in that said formed propellant powder being added as densely packed bundle of single-or multi-hole tubular propellant sticks (1) with interior combustion channels (2,3) and of large length in relation to the diameter of said combustion channels (2,3) and with a total length corresponding to at least the major fraction of the available charge space for the charge under consideration, said sticks (1) having prior thereto been provided at predetermined separations with perforations (4) from the outside of the propellant rod reaching all of its longitudinal combustion channels (2,3) said perforations (4), being effected as incisions or slits transversally through the propellant sticks (1) without the removal of any propellant material.
- 35 2. A method of producing a propellant charge for cannons as claimed in claim 1 characterized
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in that the loosely added, disoriented propellant (9) outside the bundled propellant sticks (1) is compacted to a density higher than that obtained without compression.

3. A propellant charge produced by the method according to any of the claims 1 or 2 comprising within a charge space in combination formed propellant powder (1) and loosely disoriented propellant powder which in the form of flakes, powder grains, short powder rods or tubes (9) takes up the remaining available charge space not occupied by the formed propellant powder characterized in that said formed propellant powder consists of densely packed single- or multi-hole tubular propellant sticks (1) with interior combustion channels (2,3) and of large length in relation to the diameter of said combustion channels (2,3) and with a total length corresponding to at least the major fraction of the available charge space for the charge under consideration, said sticks (1) being provided at predetermined separations with perforations (4) from the outside thereof reaching all of its longitudinal combustion channels (2,3) said perforations (4) having been effected as incisions or slits transversally through the propellant sticks (1) without the removal of any propellant material.
4. A propellant charge for cannons according to claim 3 characterized in that the loosely added, disoriented propellant (9) outside the bundled propellant sticks have been compacted to a density higher than that obtained without compression.

Patentansprüche

1. Verfahren zum Herstellen von Geschütztreibladungen für mit einer hohen Ladungsdichte und einer hohen Progressivität, die innerhalb eines Ladungsraums eine Zusammenstellung von geformtem Treibmittelpulver (1) und lose hinzugefügtem, nichtorientiertem Treibmittelpulver (11), das in der Form von Blättchen, Pulverkörnern, kurzen Ein- oder Mehrloch-Pulverstangen oder -röhren (9) den verbleibenden verfügbaren, nicht von dem geformten Treibmittelpulver besetzten Ladungsraum einnimmt, aufweisen, dadurch gekennzeichnet, daß das geformte Treibmittelpulver als dichtgepacktes Bündel von röhrenförmigen Ein- oder Mehrloch-Treibmittelstäben (1) mit inneren Verbrennungskanälen (2, 3), mit einer großen Länge im Verhältnis zum Durchmesser der Verbrennungskanäle (2, 3) und mit einer mindestens dem größten Teil des verfügbaren Ladungsraums

für die betreffende Ladung entsprechenden Gesamtlänge hinzugefügt wird, wobei die Stäbe (1) vorher in vorbestimmten Abständen mit Perforationen (4) von der Außenseite der Treibmittelstangen versehen worden sind, die alle längsverlaufenden Verbrennungskanäle (2, 3) erreichen und die als quer durch die Treibmittelstäbe (1) verlaufende Einschnitte oder Schlitze ohne das Entfernen von jeglichem Treibmittelmaterial ausgeführt sind.

2. Verfahren zum Herstellen einer Treibladung für Geschütze nach Anspruch 1, dadurch gekennzeichnet, daß das lose hinzugefügte, nicht orientierte Treibmittel (9) außerhalb der gebündelten Treibmittelstäbe (1) auf eine Dichte verdichtet ist, die höher ist als die ohne Verdichtung erhaltene Dichte.
3. Treibladung, die nach einem Verfahren entsprechend Anspruch 1 oder 2 hergestellt ist, und die innerhalb eines Ladungsraums eine Zusammenstellung von geformtem Treibmittelpulver (1) undlosem, nichtorientiertem Treibmittelpulver, das in der Form von Blättchen, Pulverkörnern, kurzen Pulverstangen oder -röhren (9) den verbleibenden verfügbaren, nicht von dem geformten Treibmittelpulver besetzten Ladungsraum einnimmt, aufweist, dadurch gekennzeichnet, daß das geformte Treibmittelpulver aus dichtgepackten, röhrenförmigen Einoder Mehrloch-Treibmittelstäben (1) mit inneren Verbrennungskanälen (2, 3), mit einer großen Länge im Verhältnis zum Durchmesser der Verbrennungskanäle (2,3) und mit einer wenigstens dem größten Teil des verfügbaren Ladungsraums für die betreffende Ladung entsprechenden Gesamtlänge besteht, wobei die Stäbe (1) in vorbestimmten Abständen mit Perforationen (4) von der Außenseite versehen sind, die alle längsverlaufenden Verbrennungskanäle (2, 3) erreichen, die als Einschnitte oder Schlitze quer durch die Treibmittelstäbe (1) ohne das Entfernen von jeglichem Treibmittelmaterial ausgeführt sind.
4. Treibladung für Geschütze nach Anspruch 3, dadurch gekennzeichnet, daß das lose hinzugefügte nichtorientierte Treibmittel (9) außerhalb der gebündelten Treibmittelstäbe auf eine Dichte verdichtet ist, die höher ist als die ohne Verdichtung erhaltene Dichte.

Revendications

1. Procédé de fabrication de charges de poudre destinées à des canons, ayant une densité élevée de charge et une progressivité élevée,

comportant, dans un espace destiné à la charge et en combinaison, une poudre (1) de poudre mise en forme et une poudre (11) non orientée et ajoutée à l'état fluide qui, sous forme de paillettes, de granulés de poudre, de courts bâtonnets ou tubes (9) de poudre à un ou plusieurs trous, occupe l'espace disponible restant de la charge qui n'est pas occupé par la poudre mise en forme, caractérisé en ce que la poudre mise en forme est ajoutée sous forme d'un groupe très tassé de baguettes tubulaires (1) de poudre à un ou plusieurs trous ayant des canaux internes (2, 3) de combustion et une grande longueur par rapport au diamètre des canaux de combustion (2, 3), la longueur totale correspondant au moins à la plus grande partie de la charge disponible pour la charge considérée, les baguettes (1) ayant été au préalable réalisées, à des distances prédéterminées de séparation, avec des perforations (4) partant de l'extérieur de la baguette de poudre et atteignant tous les canaux longitudinaux (2, 3) de combustion, les perforations (4) étant sous forme d'incisions ou de fentes disposées transversalement à travers les baguettes (1) de poudre et sans enlèvement de matière quelconque de poudre.

2. Procédé de fabrication d'une charge propulsive pour canon selon la revendication 1, caractérisée en ce que la poudre (9) non orientée, ajoutée à l'état fluide à l'extérieur des baguettes (1) de poudre qui sont groupées est tassée à une densité supérieure à celle qui est obtenue sans compression.

3. Charge de poudre produite par le procédé selon l'une des revendications 1 et 2, comportant, dans un espace destiné à une charge, une combinaison formée d'une poudre propulsive (1) et d'une poudre non orientée et fluide, qui est sous forme de paillettes, de granulés de poudre, de courts bâtonnets ou tubes de poudre (9) occupant le reste de l'espace disponible pour la charge et qui n'est pas occupé par la poudre mise en forme, caractérisée en ce que la poudre mise en forme est constituée de baguettes tubulaires (1) de poudre bien tassée, ayant un ou plusieurs trous, avec des canaux internes (2, 3) de combustion et ayant une grande longueur par rapport au diamètre des canaux de combustion (2, 3), leur longueur totale correspondant au moins à la plus grande partie de l'espace disponible pour la charge considérée, les baguettes (1) ayant, à des intervalles prédéterminés de séparation, des perforations (4) partant de l'extérieur et atteignant tous les canaux longitudinaux de combustion

5 (2, 3), les perforations (4) étant réalisées sous forme d'incisions ou de fentes disposées transversalement à travers des baguettes (1) de poudre sans enlèvement de matière de poudre.

10 4. Charge de poudre pour canon selon la revendication 3, caractérisée en ce que la poudre (9) non orientée qui est ajoutée à l'état fluide à l'extérieur des baguettes regroupées de poudre a été comprimée à une densité supérieure à celle qui est obtenue sans compression.

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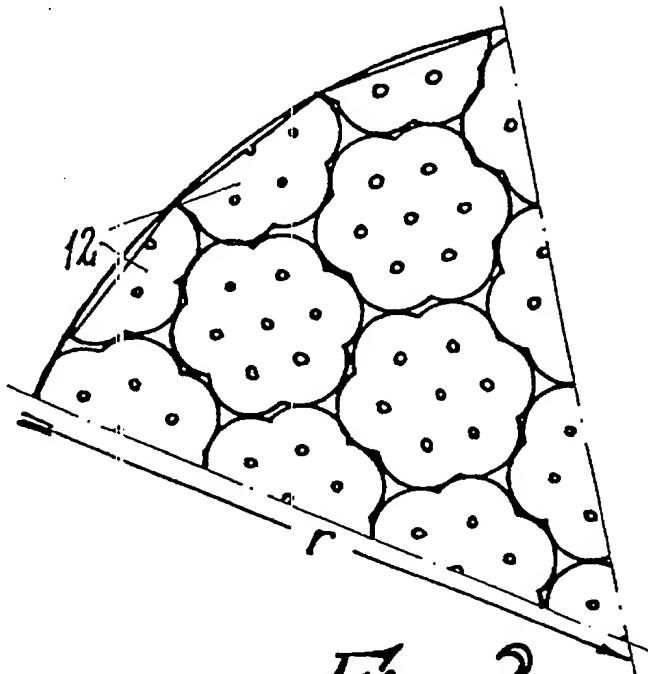


Fig. 3

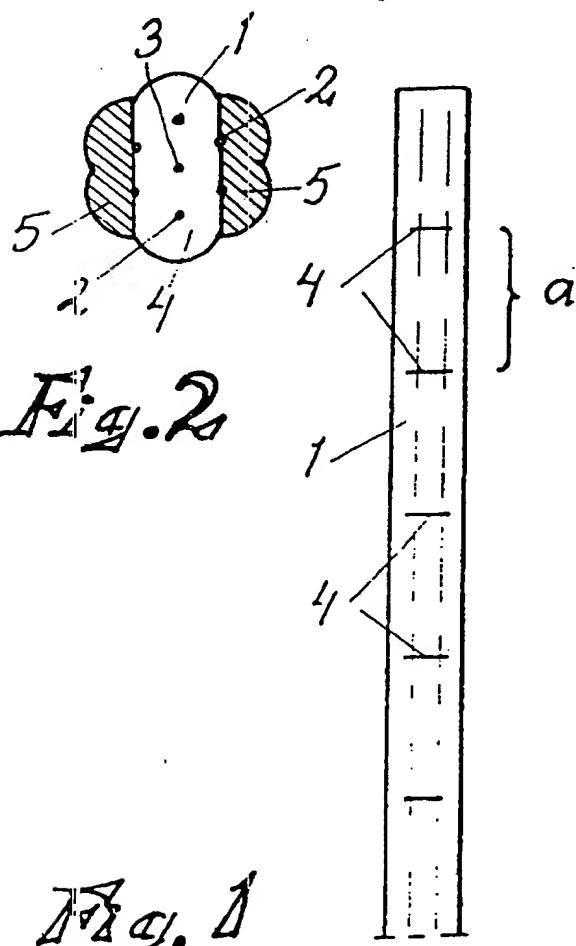


Fig. 2

Fig. 1

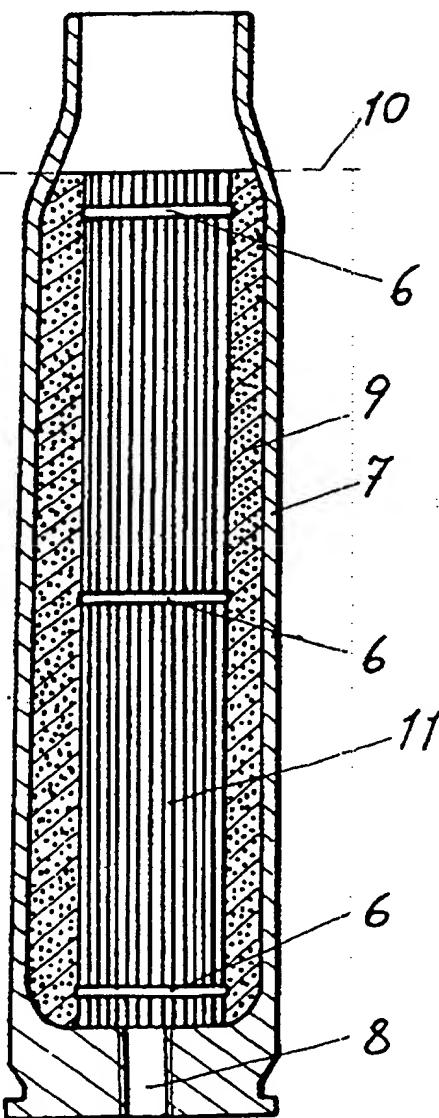


Fig. 4